

Global Precipitation Measurement

*System Requirements Review
System Architecture & Concept*

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- *Driving Level 1 Requirements*
- *Architecture Overview*
- *Trades Performed*
- *Functional Allocation*
- *Operations Concept*
- *Remaining Trades*
- *Technologies*

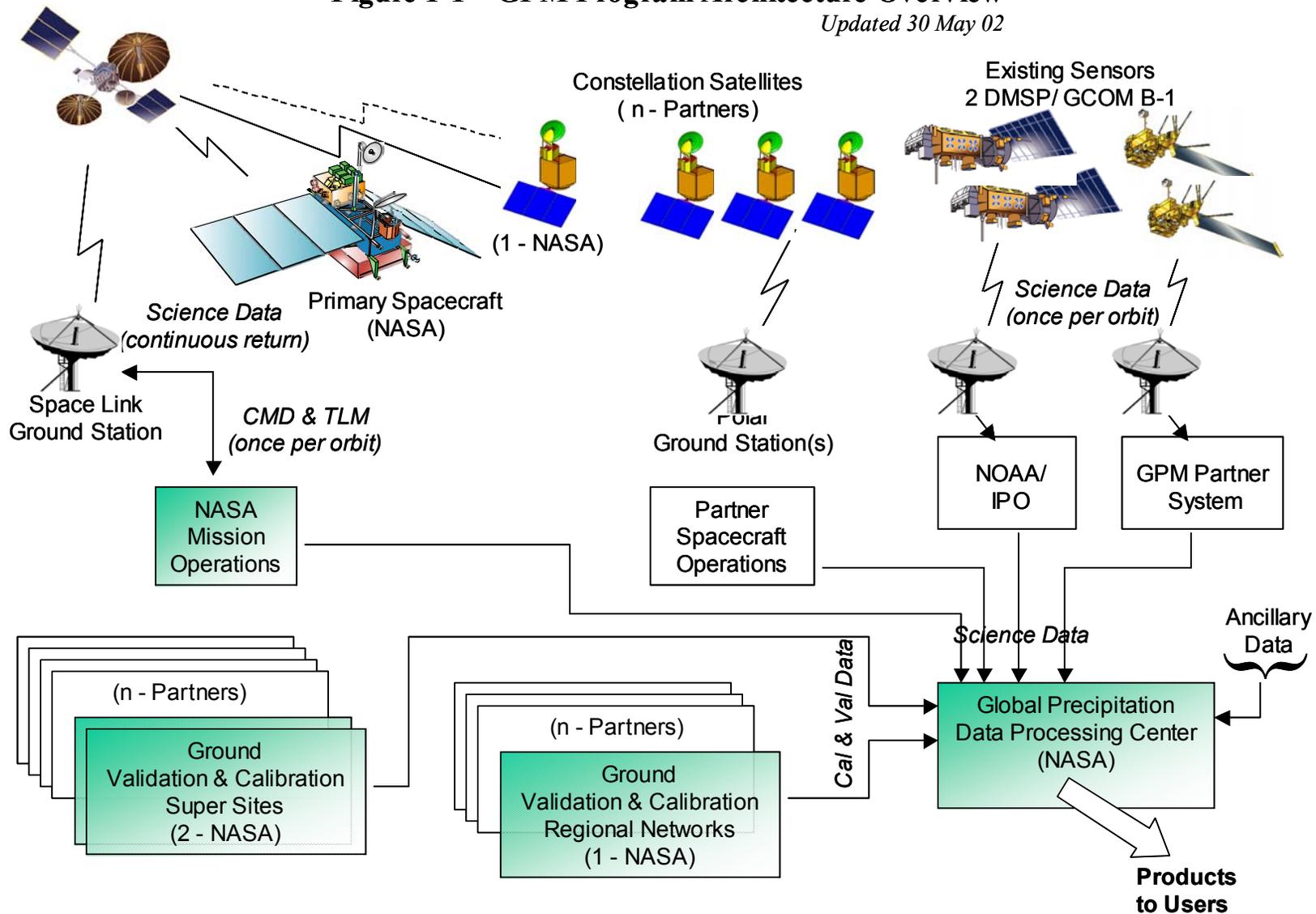


- **Precipitation sampling**
 - Resolve diurnal cycle
 - Ingest multiple data streams from NASA and domestic and foreign partners
- **Drop size distribution measurement**
 - Fundamental feature of precipitation
 - Enabled by dual-frequency precipitation radar (DPR)
 - 5km horizontal and 500m vertical resolution
- **Error estimates**
 - Critical to users
 - Drives GV and algorithm efforts
- **NASA/NASDA partnership**
 - DPR
 - H2A-202 launch vehicle
- **Latency and Completeness**
 - 3 hours for 90% of data
 - 72 hours for 98% of data
- **Lifetime: 3 years with a goal of 5**
- **End-of-Life disposal**



Figure 1-1 GPM Program Architecture Overview

Updated 30 May 02



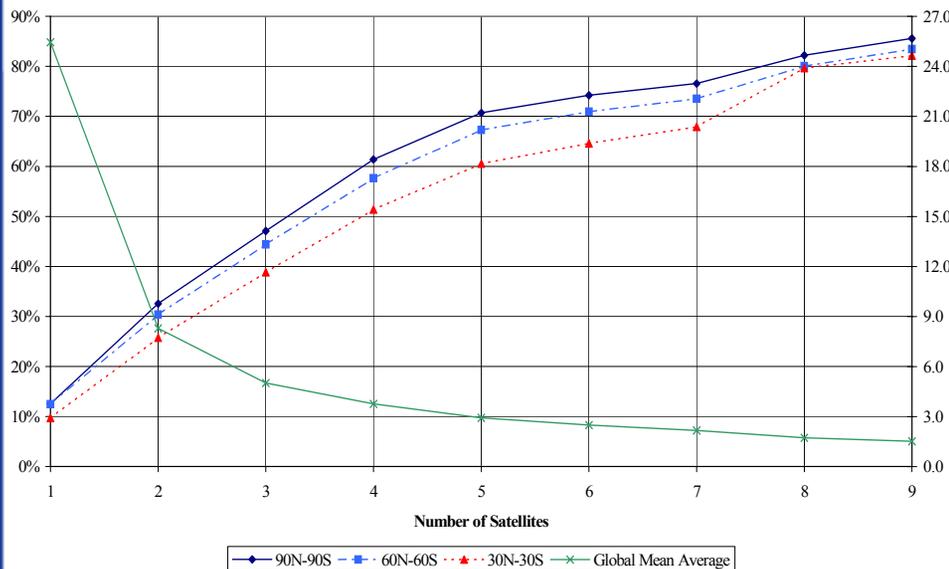
- *Orbit Studies*
- *Current CCSDS Standards vs. Internet Protocol*
- *Ground Network vs. TDRSS*
- *Direct Broadcast Options*
- *Two TDRSS Antennas vs. GN Backup*
- *Constellation Configuration*
- *Other Trades*



- **Constellation orbit analysis**

- Revisit time
- Binning statistics
- % coverage in 3 hours

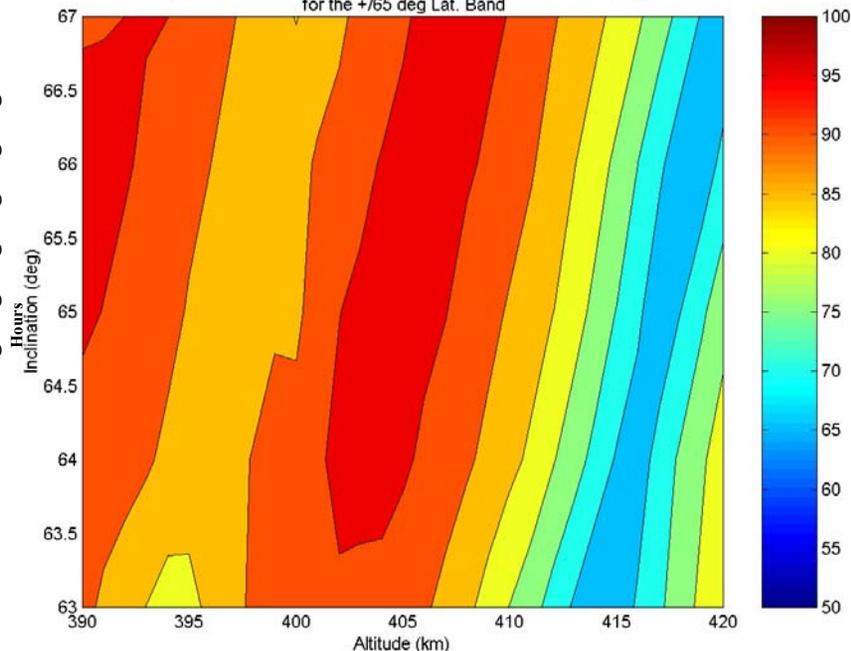
Percent of 3-hr bins Sampled and Global Mean Revisit Time



- **Core orbit analysis**

- Drag make-up
- Time to cover total area

Coverage % Achieved by the KU Band Radar After 1 Week Propagation for the +/-65 deg Lat. Band



• **More details to follow in Chad Mendelsohn's Presentation**



- *Internet Protocol (IP) is an international standard with tremendous commercial technology investment.*
- *NASA and the Consultative Committee for Space Data Systems (CCSDS) are working toward the use of standard commercial interface standards.*
- *With a launch in 2007 and operations continuing past 2010, GPM must look toward compatibility with future missions, not past missions.*
- *IP has been successfully demonstrated between orbiting spacecraft and the ground.*
- *GPM will use Internet Protocol for the Space-to-Ground link.*
- *Steve Tompkins will discuss more details and implications of IP.*



- *Five stations required for core spacecraft to meet 3 hour latency due to 65° inclination*
- *GN solution may be practical for polar orbiter*
- *Tim Rykowski will present more information on this trade during the Mission Operations System presentation.*



- ***Compared Direct Broadcast Transmitter to using TDRSS Demand Access Service Continuously***
 - *Direct broadcast requires extra flight hardware*
 - *Direct broadcast only available to users in view of spacecraft*
 - *Direct broadcast requires special ground equipment to receive*
 - *TDRSS DAS allows us to get science data to the PPS with very little latency – improves the 3-hour products*
- ***Selected TDRSS Demand Access Service***



- **Advantages of GN Backup**
 - *Spacecraft will fly omni antennas anyway*
 - *Significant cost associated with second antenna*
 - *Drag*
 - *cp/cg offset*
 - *Hardware costs*
 - *Integration and testing costs*
 - *Extra mass*
 - *GN cost not incurred unless failure occurs, but some data will be lost while GN contracts are established*
 - *Antenna is fairly high reliability – low probability of incurring GN costs*
- **Advantages of two TDRSS antennas**
 - *Two HGA's would enable make before break handovers*
 - *GN solution is costly*
 - *GN solution makes meeting 3 hour latency requirement difficult*
- **Baselined a single TDRSS antenna on the primary spacecraft, with GN used as backup.**



- ***Orthodox constellation vs. collection of existing data streams***
 - *Cost of launching nine spacecraft into different orbital planes is prohibitive*
 - *Low data volume and advancing computer and network capability and standardization makes sensor fusion from partners practical*
 - *TRMM Data System capabilities are being extended to demonstrate this concept*
- ***GPM will use a collection of data streams from various spacecraft microwave radiometers, with NASA providing only two spacecraft.***

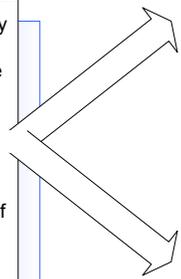


- *Element trades to be described tomorrow*



GPM Science Drivers

- * **Climate Observation** - to accurately measure global/regional variability of precipitation to improve global climate models.
- * **Weather Prediction** - to improve the accuracy of global and regional weather prediction models.
- * **Global Water Cycle** - to improve the understanding and predictability of relevant components of the water cycle.



SPACE SEGMENT:

- * **Measurement:**
- * **Global Coverage** - instantaneous rain rate measurements sufficiently frequent to avoid diurnal aliasing, and with 10 km resolution.
- * **Measure** 4-D structure of rainfall rates and drop size distribution.

Primary Spacecraft

- * Carry a multi-instrument payload
- * Obtain Precipitation Measurements
- * Obtain Latent Heat Release Measurements
- * Obtain Drop Size Distribution Measurements

Constellation Spacecraft

Dedicated Spacecraft

- * Carry a dedicated passive microwave radiometer
- * Obtain Precipitation Measurements
- * Obtain Global Coverage
- * Obtain Temporal Coverage

Partner Spacecraft

- * Carry a suitable passive microwave radiometer
- * Obtain Precipitation Measurements
- * Contribute to Global Coverage
- * Contribute to Temporal Coverage

Mission Operations Center

- * Manage NASA Space Assets
- * Co-ordinate NASA/ Partner S/C Operations
- * Co-ordinate S/C- Ground Site Overflights
- * Co-ordinate Primary-Constellation Coincident Observations

GROUND SEGMENT:

- * **Validation:**
- * **Accuracy Threshold** - Bias error < 5%
- * **Precision Threshold** - < 25%
- * **Science Products:**
- * **Latency** - Deliver near-real-time and 3-hour products.
- * **Climate Products** -
- * **Outreach Data Access** -

Ground Validation & Calibration

Super Site

- * Collect Precipitation, Cloud and DSD data
- * Obtain atmospheric temperature and moisture profile measurements
- * Obtain radiation, heat and water fluxes
- * Develop data quality estimates
- * Develop error characterization data

Regional Network

- * Collect Precipitation Data over a scale greater than 250 km.
- * Develop data quality estimates

Precipitation Data Processing Center

- * Generate Science Products
- * Provide Outreach Data Access
- * Develop Cal/ Val Parameters



- **Mission-level operations concept describes high-level interaction among elements**
 - Primary spacecraft (also called Core Spacecraft)
 - Constellation spacecraft
 - Mission Operations System
 - Precipitation Processing System (PPS)
 - Ground Validation System
- **Primary Spacecraft**
 - Carries DPR and GMI
 - Provides reference standard for constellation
 - Makes DSD and latent heat measurements
 - Launches on H2A-202
 - Downlinks science data continuously via TDRSS-MA
 - Uses IP for space-to-ground link
 - On-board MIL-STD-1553 for instrument interface, Ethernet under study
 - Autonomous navigation for orbit control
 - Design for demise



- **Constellation Spacecraft**
 - Carries only a radiometer (copy of GMI)
 - Enhances coverage provided by partners \Rightarrow improves sampling
 - Uses TDRSS-MA continuously
- **Mission Operations System**
 - Controls the spacecraft
- **Precipitation Processing System**
 - Ingests data streams from core, constellation, and partner spacecraft
 - Ingests data from GV sites
 - Ingests ancillary data (such as IR)
 - Sends overpass info to GV sites
 - Output products
 - Outreach rain map via web server
 - 3-hour products
 - Climate products
- **Ground Validation System**
 - Collects data and produces algorithms needed to characterize errors
 - Provides products to PPS on a regular basis
 - Runs targeted campaigns to resolve specific issues in the precipitation retrieval algorithms

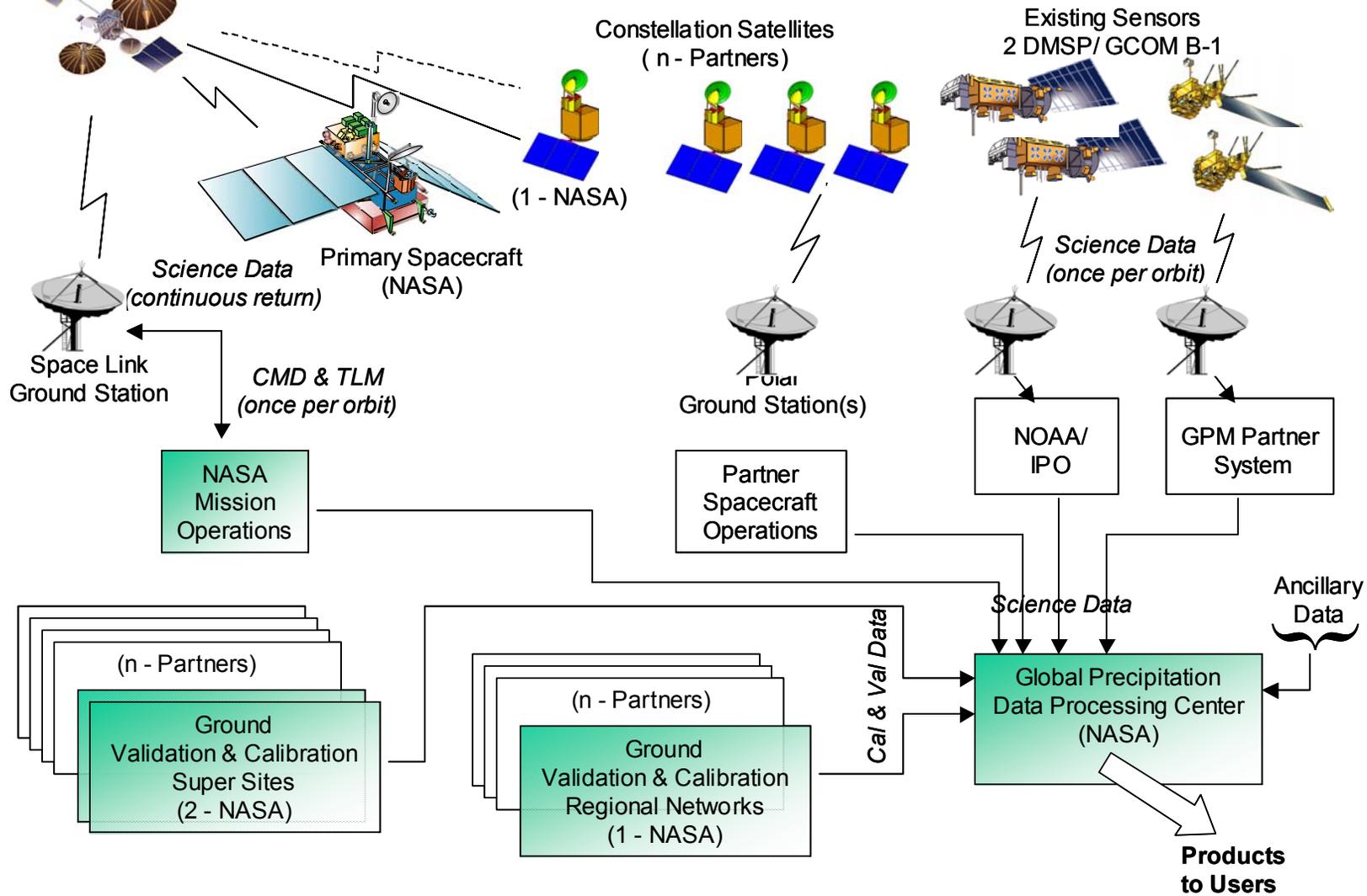


- *More detailed operation of each element will be described by element engineers*
- *Draft operations concept document on DocuShare*

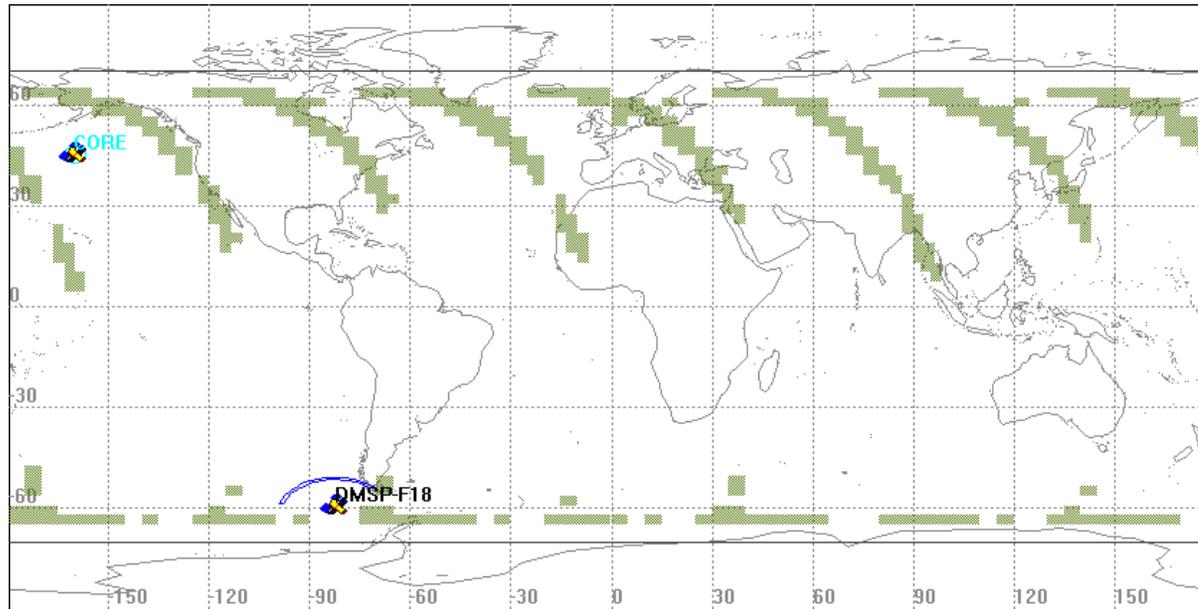


Figure 1-1 GPM Program Architecture Overview

Updated 30 May 02



- *Using core/constellation crossings vs. using GV over-flights to calibrate across the constellation*
 - *Core/constellation crossings are infrequent*



One month
of crossings

- *TRMM has shown that absolute accuracy of GV measurements is difficult to achieve*
- *Finalizing constellation spacecraft orbit*
- *Finalizing constellation spacecraft launch vehicle*
- *Many trades at the element level*

- ***GPM will utilize technology, not develop it.***
 - *Functional capability by PDR, design specific by CDR*
 - *Standard techniques carried in design until new technology is ready*
- ***Autonomous navigation***
 - *Demonstrated by EO-1*
 - *Provides finer orbit control*
 - *Reduces operations cost*
- ***Design for demise***
 - *Reduces fuel by 40%*
 - *Inherently safer*
- ***Internet Protocol***
 - *Demonstrated on UOSAT-12 by OMNI*
 - *International standard with tremendous commercial investment in technology*
 - *Simplifies autonomous data delivery*
- ***Additional technologies under consideration by element managers***

